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NATION AL **AERONAUTICS**

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AN ANALYSIS FOINT-TO-FOINT COMMUNICATION

APPLICATION TO THE LUNAR FLYER PROGRAM

(THRU)

(CODE)

(CATEGORY)



MANNED HOUSTON, TEXAS SPACECRAFT CENTER

REPRODUCIBILITY PAGE IS POOR.

MSC INTERNAL NOTE MSC-EE-K-68-15

AN ANALYSIS OF POINT-TO-POINT COMMUNICATION

APPLICATION TO THE LUNAR FLYER PROGRAM

Appendix by Jeffelson F. Lindsey III Approved by J. Pawlowski Section Approved by J. Chicolica Approved by L. Chicolica Approved by L. Chicolica	E. L. Chicoine, Chief Electromagnetic Systems Branch		Branch
Appendix by J. Approved by Ant.	Approved by & L. Chier ins	Approved	
Appendix by J. Pawlowski Approved by 1. Sales			Section
Appendix by Jeffelson F. Lindsey III J. Pawlowski	Approved by to sepen	Approved	
SHE Y	Appendix by J. Pawlowski	Append1x	
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NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
Manned Spacecraft Center
Instrumentation and Electronic Systems Division
August 1968

of-sight horizon on the lunar surface. For this analysis, the transmitter output is assumed to be 1 watt (+30 dbm) and the antenna heights are frequency for point-to-point communication in the vicinity of the line-This study is concerned with an analysis to determine the optimum

sensitivities. Each parameter is briefly explained sis include the free-space transmission loss, the Bremmer Series loss, the ground proximity loss, antenna gains, transmitter power and receiver The circuit margin parameters used in the communication link analy-

power received to the power transmitted assuming two isotropic antennas The free-space transmission loss is defined as the ratio of the

$$L_o = \frac{\lambda^2}{(4\pi d)^2}$$
 (1)

diffraction by the lunar surface. The Bremmer Series loss is defined as Series loss represents the loss which occurs as a result of blockage and λ = the wavelength and d = distance between antennas.

$$L_{B} = 8\pi J \left| \sum_{n=1}^{\infty} \frac{e^{-\tau_{n}J}}{8'' + 2\tau_{n}} f_{n}(h_{1}) f_{n}(h_{2}) \right|^{2}$$
 (2)

Series and free-space losses. KM and 30 KM. A more detailed explanation of the Bremmer loss is given in quencies ranging between 200 KHz and 2200 MHz and distances between 2.5 with the Bremmer Series diffraction loss is tabulated in Table 1 for freparameter and $f_n(h)$ is the height-gain function. The free-space loss along I is a distance function, " is the mode number, S" is a ground A computer analysis was used to determine both the Bremmer The first seven terms of the Bremmer Series

in the Appendix. son by Mr. Jim Pawlowski of the Digital Techniques Section and is included The program was formulated under the direction of Mr. Ray Thomp-

independent of the moon's dielectric parameters. quencies; whereas, at VHF and above the transmission loss is practically dielectric parameters of the moon are more significant at the lower frecreases the transmission loss becomes smaller. Also it is noted that the a function of frequency and range. It is noted that as the frequency dea plot is given in Figure 1 depicting the changes in transmission loss as this is especially true at lower frequencies. that higher conductivity contributes to a longer communication range and A tabulation of the total transmission loss is shown in Table II and In general, it is found

when the antenna is electrically close to the lunar surface. or that loss which affects the radiation resistance of an antenna because in Table III are typical antenna gains for the frequencies involved. seen in Table III, this loss becomes important only at low frequencies graphs given by (Vogler, 1964) and are tabulated in Table III. of its close proximity to the ground. and the ground proximity loss. the right side of Table III is listed the combined effects of antenna gains The next factor to be considered is that of the ground proximity loss These losses have been taken from Also shown As may be

with a received signal-to-noise ratio of 18 db. Also, shown in Table IV is the allowed antenna port-to-antenna port loss which includes +30 dBm sitivity which is shown in Table IV. An RF bandwidth of 40 KHz is assumed (1 watt) for the transmitter and the assumed receiver sensitivities at The final frequency variable factor to be considered is receiver sen-

the various frequencies.

plotted in Figure 2 vision bandwidths, frequency of 10 MHz with a data bandwidth of 14 KHz (RF bandwidth of astronaut frequency band of and from the combined effects tabulation in Table With the assumption of a would result an output signal-to-noise ratio of 18 db. may go to a range of 10 kilometers the 10 MHz carrier frequency would 88 VHF or higher would be required and a shorter maximum a function of frequency. Н Watt (30 dbm) transmitter one may deterand still For V and Table IV that the The maximum range is application with teleton communicate be sufficient,

could naut could carry a low power receiver and a direction finding antenna with pending on the lunar surface conductivity. and assuming a 10 db less efficient direction finding antenna. gain of be used on the Lunar Module yielding a approximately direction finding, the 10 MHz frequency could -15 db with respect 20 kilometers to isotropic. by using a small NV bandwidth of 1 KHz 'A 1 watt range of 20-23 kilometers de-10 MHz transmitter e d used out The astro-

9 42 the conductivity of the lunar surface. foot antenna heights is found to be in the 1 The optimum frequency for point-to-point communication with ф О 10 MHz range depending O١ foot and

This analysis very large compared selection. has not communication system this ó considered antenna size. AHV antennas of equivalent factor could Low frequency antennas characteristics weigh heavily

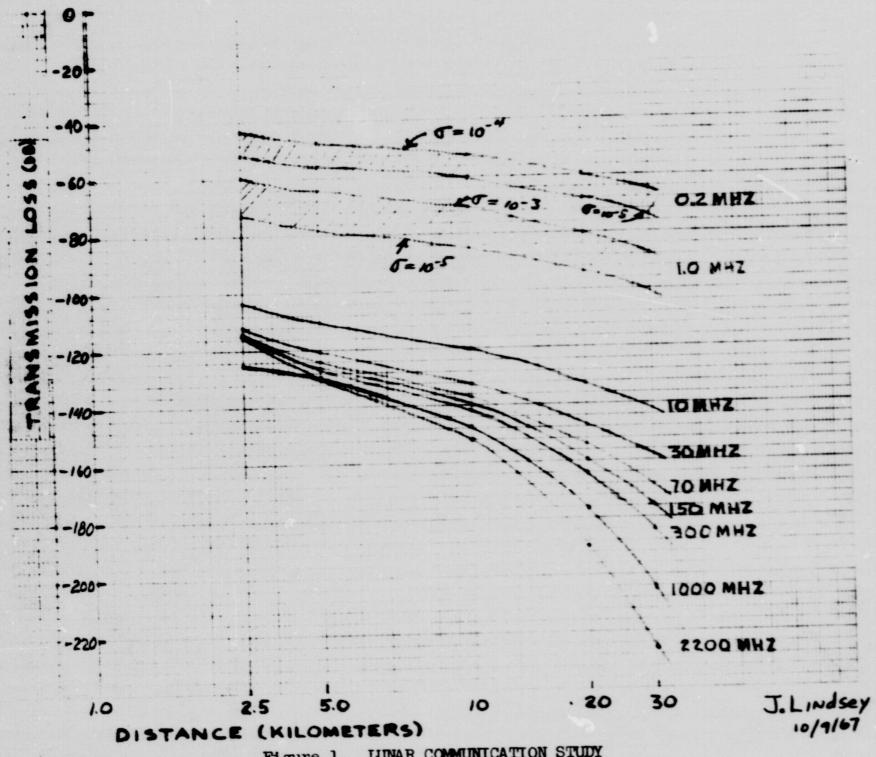


Figure 1. LUNAR COMMUNICATION STUDY

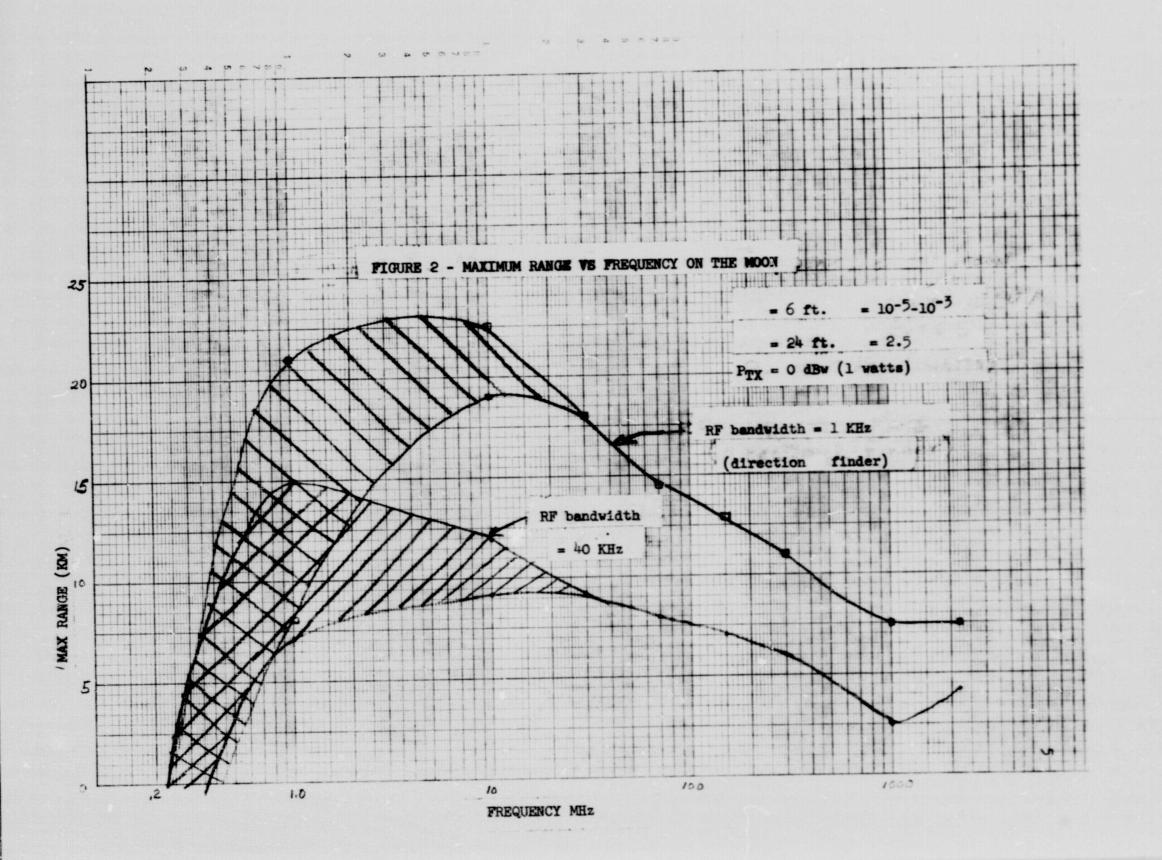


TABLE I - FREE SPACE AND BREMMER LOSSES

TABLE I (cont.)

0.0	0	-	-	1	10	10	OT	30	30	30	70	70	70	150	150	150	300	300	300	T000	1000	1000	2200	2200	2200	(MHz	Freq
10-4	10-5	10-3	10-4	10-5	10-3	10-4	10-5	10-3	10-4	TO-2	10 - 3	10-4	10 - 5	10 - 3	10-4	70 - 5	10-3	10-14	10 - 5	10-3	10-4	10-5	10 - 3	10-4	10 - 5	J/M	Freq Conductivity
-38.5	- 38.5	-52.4	-52.4	-52.4	-72.4	-72.4	-72.4	-81.9	-81.9	-81.9	-88.8	-88.8	-88.8	-95.9	-95.9	-95.9	-101.9	-101.9	-101.9	-112.4	-I12.4	-112.4	-119.2	-119.2	-119.2	Free Space Loss(db)	у 10.0
-13.8	-22.7	-18.4	-32.5	-32.2	-48.7	4.74-	4.74-	-50.7	-50.3	-50.3	-48.6	-48.6	-48.6	-44.6	-44.6	-44.6	-40.6	-40.6	-40.6	-34.9	-34.9	-34.9	-33	-33	-33	Bremmer Loss (db)	Kim
-44-5	-44.5	-58.4	-58.4	-58.4	-78.4	-78.4	-78.4	-87.9	-87.9	-87.9	-94.8	-94.8	-94.8	-101.9	-101.9	-101.9	-107.9	-107.9	-107.9	-118.4	-118.4	-118.4	-125.2	-125.2	-125.2	Free Space Loss(db)	20.0
-14.7	-23.8	-21.9	-35.9	-35.6	-56.0	-54.7	-54.7	-59.6	-59.3	-59.3	-59.3	-59.3	-59.3	-57.9	-57.9	-57.9	-56.9	-56.9	-56.9	-59.0	-59.0	-59.0	-64.7	-64.7	-64.7	Brenmer Loss(db)	香
_48.0	-48.0	-61.9	-61.9	-61.9	-81.9	-81.9	-81.9	-91.4	-91.4	-91.4	-99.3	-99.3	-99.3	-105.4	-105.4	-105.4	-111.4	-111.h	-111.4	-121.9	-121.9	-121.9	-128.7	-128.7	-128.7	Free Space Loss (db)	0
-16.8	-25.7	-25.6	-39.6	-39.3	-61.7	-60.5	-60.5	-67.3	-67.0	-67.0	-69.3	-69.3	-69.3	-70.6	-70.6	-70.6	-72.9	-72.9	-73.0	-83.7	-83.7	-83.7	-97.5	-97.5	-97.5	Loss (db)	Kim
						-			-	-							,	<u> </u>									Type of

TABLE II. TOTAL TRANSMISSION LOSS

FREQ	6	2.5 KM	5.0 KM	10 KM	20 KM	30 KM
(MHz)	16 7M	Trans. Loss	Trans.Loss (db)	Trans.Loss (db)	Trans.Loss (db)	Trans. Loss (db)
2200	10-5	-115.2	-130.4	-152.2	-189.7	-226.2
2200	10-4	-115.2	-130.4	-152.2	-189.7	-226.2
2200	10-3	-115.2	-130.4	-152.2	-189.7	-226.2
1000	10-5	-124.9	-128.8	-147.3	-177.4	-205.6
1000	10-4	-124.9	-128.8	-147.3	-177.4	-205.6
1000	10-3	-124.9	-128.8	-147.3	-177.4	-205.6
300	10-5	-115.0	-127.2	-142.5	-164.8	-184.4
300	10-4	-115.0	-127.2	-142.5	-164.8	-184.4
300	10-3	-115.0	-127.3	-142.5	-164.8	-184.4
150	10-5	-115.4	-126.4	-140.5	-159.8	-176.0
150	10-4	-115.4	-126.4	-140.5	-159.8	-176.C
150	10-3	-115.4	-126.4	-140.5	-159.8	-176.0
70	10-5	-114.7	-124.3	-137.4	-154.1	-168.6
70	10-4	-114.7	-124.3	-137.4	-154.1	-168.6
70	10-3	-114.7	-124.3	-137.4	-154.1	-168.6
30	10-5	-112.2	-120.5	-132.2	-147.2	-158.4
30	10-4	-112.2	-120.5	-132.2	-147.2	-158.4
30	10-3	-112.6	-120.8	-132.6	-147.5	-158.7
10	10-5	-103.2	-110.0	-119.8	-133.1	-142.4
10	10-4	-103.2	-110.0	-119.8	-133.1	-142.4
10	10-3	-104.4	-111.3	-121.1	-134.4	-146.3
1	10-5	-73.2	-78.1	-84.6	-94.0	-101.2
1	10-4	-73.5	-78.3	-84.9	-94.0	-101.
1	10-3	-59.4	-64.3	-70.8	-80.3	-87.5
0.2	10-5	-51.9	-56.0	-61.2	-68.3	-73.7
0.2	10-4	-43.0	-47.0	-52.3	-59.2	-64.8

00

Cr = 2.5

Grand PROX. Loss

Antenna

Gain

		010000		2000	Coccon	
		Pag-	L _{p24}	90	G ₂₄	Total Effect of
f(Mz)	σ(ω/w)	61	24.	05T 8Q	150 BQ	B
2200	10-5	0	0	t-	-2	-6
2200	10-4	0	0	-4-	-2	- 6
2200	10-3	0	0	μ-	-2	-6
1000	TO-2	0	0	-5	-2	-7
1000	10-4	0	0	-5	-2	-7
T000	10-3	0	0	-5	-2	-7
300	C-01	0	0	-5	-2-	-7
300	10-4	0	0	-5	-2	-7
300	10-3	0	0	- 5	-2	-7
150		0	0	-4-	-1	- 5
150	10-4	0	0	-4-	-1	-5
150	10-3	0	0	-4	-1	- 5
70	TO2	0	0	-3	0	-3
70	TO-4	0	0	- 3	0	.
70	10-3	0	0	- 3	0	ئ
30	70-2	-1	0	-4	0	-5
30	10-4	-1	0	-4	0	-5
30	10-3	-1	0	-4	0	-5
0.0	TO-2	-2	0	- 5	-5	-12
10	TO-4	-3	0	- 5	-5	-13
10	10-3	-5	0	- 5	- 5	-15
1	TO-2	-23	-10	-10	-10	-53
1	10-4	-30	-17	-10	-10	-67
1	10-3	-26	-13	-10	-10	-59
.2	70-5	-53	- 35	-17	-17	-112
.2	10-4	-48	-35	-17	-17	-117
.2	10-3	-43	-25	-17	-17	-102

L = Ground Proximity Loss
G = Antenna gain plus cable losses

		ALLOWED LOSS	LOSS
Freq. (MHz)	Srx (dbm)'	RFBW = 40 KHz	RFBW = 1 KHz
.2	-67	97	106
1	-103	133	149
10	-104	42τ	150
30	-105	135	151
70	- 501-	135	151
150	-105	135	151
300	-105	135	151
1000	-101	151.	747
2200	-101	1.51	747

^{&#}x27;S/No = 18 db \$ R.F. Bw = 40 KHZ

TABLE IV - RECEIVER SENSITIVITIES

And Allowed Antenna port-to-antenna port

loss for PTX = 30 dbm

^{&#}x27;Obtained from A. Pajak and J. Fowler

1 10-3 1 10-3	1 10-3	1 10	-4	Z-01 T	10 10 ⁻³	10 10 ⁻⁴	2-ot ot	30 10 -5-10-3		12-5-01 05T	300 TO_2-TO	1000 10-5-10	2200 10-5-10-3	Freq. Conductivi		
- TOO. 7	-163 0	-118.4	-140.5	-126.2	-119.4	-116.2	-115.2	0-3 -117.2	5_10-3 -117.7	-10-3 -120.4	-10 ⁻³ -122.0	-10 ⁻³ -131.9	-121.2	ivi. 2.5	Die	
STREET, STREET	-168.0	-123.3	-145.3	-131.1	-126.3	-123.0	-122.0	-125.5	-127.3	-131.4	-134.2	-135.8	-136.4	5.0	Distance(KM)	
	-173.2	-129.8	-151.9	-137.6	-136.1	-132.8	-131.8	-137.2	-140.4	-145.5	-149.5	-154.3	-158.2	0.01		
	-180.2	-139.3	-161.0	-147.0	-149.4	-146.1	-145.1	-152.2	-157.1	-164.8	-171.8	-184.4	-195.7	20.0		
	-185.7	-146.5	-168.2	-154.2	-158.6	-155.4	-154.4	-163.4	-171.6	-181.0	-191.4	-212.6	-232.2	30.0		

TABLE V. Transmission Loss, Antenna Gains and Ground Proximity Loss

References

- Burrows, C. R.: Radio Wave Propagation. Academic Press, Inc. 1949.
- $\dot{\omega}$ Volger, L. E.: A Study of Lunar Surface Communication. NBS Monograph
- 85, September 14, 1964.

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APPENDIX

by J. Pawlowski

Computer Programs

for

Equations (1) and (2)

Analysis of Free Space Loss

The general formula to calculate Free Space Loss expressed in decibels is

$$L_0(ab) = 10 \log_{10} \frac{|\hat{E}|^2}{|E|^2} = 10 \log_{10} \frac{\lambda^2}{(4\pi d)^2}$$

 $\mathbf{E}_{o} = \mathbf{received signal}$

 $\hat{\mathbf{E}}_{s}$ = transmitted signal

 λ = wave length

d = distance between antennas

H is necessary to express λ and d in the same units:

Inerefore, the speed of light. Hence, it and d must be expressed in the it can be seen that is the same to say that the same distance units. the distance units of λ are the same as those speed of light O H

Likewise,

$$\lambda(f) = \frac{984}{\text{frequency (MHZ)}}$$

When d is expressed in meters, use λ (m), and when d is expressed in use $\lambda(f)$. If it is necessary to express d in other units adjust λ expressing the speed of light in the same units. feet

depends In order to include the loss and/or gain due to reflection and refraction we must compute the Geo-Optics or the Bremmer Series methods and add these results to the Free Space Loss. The choice of what method to use depends on the distance between the antennas.

If the distance between the antennas is greater than 25% of the line of sight then the Bremmer Series is used. If it is less than or equal to this value the Geo-Optic method is used.

Line of Sight = $\sqrt{2kah_1}$ + $\sqrt{2kah_2}$

a = radius of the body

= index of refraction

 $_1$ = height of one antenna

 h_2 = height of the other antenna

CAUTION! a, h_1 , and h_2 must be expressed in the same units of distance.

FREE SPACE LOSS

OPERATION OF PROGRAM

- H Load the Fortran operating system using the migh speed reader.
- io and Put turn the Free Space Loss ဓ္ဓ the reader. Program object tape in the low speed reader
- ω Load address 200 and depress start the tape will load.
- + When the tape stops, depress the continue switch. The message "Type Frequency" will be typed out on the teletype. Enter the frequency expressed in MHZ on the keyboard. Express all entries in Floating Point except the number of increments (Step 9). The message "Type
- VЛ expressed in mega-units. "Type Speed of Light" is then typed. Enter the speed of light
- 9 index of refraction" is the next message. Enter this quantity.
- 7 "Type radius of Body" and "Type heights of antennas" are the next to messages type out. Remember that these quantities must be entered using the same units of distance. messages type out. using the same uni the next two
- φ. The value for λ , the wave length, is calculated and typed out
- 9. The message, "Type the number of increments", is typed. Enter in fixed point the total number of antenna distances that you wish to investigate.
- 10. number +1 which labels the first distance will then be typed.
- 11. The message "Type the distance between antennas" is typed. to enter this in the same units as the speed of light. Remember
- . 12 The "Free Space Loss" db are typed out. followed by the "Free Space Loss" expressed in
- 13. The message whether to use Geo Optic next calculations is typed. typed. ೧೭ use Bremmer series for the
- 14. Then the number +3, etc. number +2 is typed and the steps 11 through 13 are repeated

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ACCEPT 8, A
ACCEPT 8, A
ACCEPT 8, A
ACCEPT 8, A
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EXPLANATION OF PROGRAM

The program is broken into three sections:

is solved in this section on the 11th line after 9;. accept data. This section extends from the line Labeled 1; through two lines after the line labeled 12;. However, the equation for wave length Section I is composed almost entirely of commands to type messages and

corresponds to

Section II contains the calculations and outputs for the Free Space Loss and extends from the end of Section I to the second line after 14;

$$FSL = (AMBA/(4.*3.1416*D))**2$$

corresponds to

$$L_0 = \left(\frac{\lambda}{4\pi d}\right)^2$$

$$DB = 4.31,294*LOGF (FSL)$$

corresponds to

$$L_o$$
 (DB) = 10 X Log_{10} (L_o)

The factor of .434294 is used to equate the LOGF, which is \log_e , to \log_{10} .

method should next be Section III contains the test to see if the Bremmer Series or Geo-Optic used.

corresponds to

DIST = DIST + SQTF (2.*REFR*A*H2)

corresponds to

Line of Sight =
$$\sqrt{2 \text{kah}_1} + \sqrt{2 \text{kah}_2}$$

DIST = 0.25*DIST

is 25% of the line of sight.

Next is the "IF" statement which determines whether 25% of the line of sight is greater than the distance between antennas or less than or equal to the distance between antennas.

The general equation for the Bremmer Series expressed in decibels is

$$L_{\rm B}({\rm dib}) = 10 \log_{10} \frac{|\hat{\mathbb{H}}|^2}{|\hat{\mathbb{H}}|^2} = 10 \log_{10} \left[2(2\pi f)^{\frac{1}{2}} \left| \sum_{n=1}^{7} \frac{e^{-j n} f}{\sqrt{+27n}} f(h_2) \right| \right]^2$$

This can be simplified to

$$\hat{\mathbf{H}} = \text{received field} \left[8\pi S \cdot |f(h_i)|^2 \cdot |f(h_i)|^2 \cdot |\sum_{h=1}^{7} \frac{e^{-j\pi_h S}}{S + 2\pi_h} \right]$$

= Free Space Field

distance factor

ground parameter

= heights of antennas

= complex numbers which characterize the individual terms in equation (1)

For vertical polarization where
$$i = 1, 2$$

$$f(h_i) = i + j \left(\frac{2\pi h_i}{\lambda} \sqrt{\frac{\epsilon_{c-1}}{\epsilon_{c}}}\right)$$

hi~ 302 33

For horizontal polarization where i = 1, $f(h_i) = 1 + j \left(\frac{2\pi h_i}{\lambda} \sqrt{\epsilon_{c}-1}\right)$

hi < 3012/3

$$\hat{\epsilon}_{e} = \epsilon - j (60 \lambda \sigma)$$
 $\hat{\epsilon}_{e} - l = (\epsilon - l) - j (60 \lambda \sigma)$
 $\hat{\epsilon}_{e} - l = \sqrt{(\epsilon - l)^{2} + (60 \lambda \sigma)^{2}}$

j tan (-6010)

ddelectric constant of the spherical body

wave length of signal

conductivity of the spherical body

If
$$W = 60 \lambda O$$
Then $\sqrt{\epsilon_{c}-1} = (\sqrt{(\epsilon-1)^{2} + W^{2}})^{\frac{1}{2}}$, $e^{\int_{0}^{\infty} \frac{tan^{-1}(\frac{-60\lambda O}{\epsilon-1})}{2}}$

If $TINY = (\sqrt{(\epsilon-1)^{2} + W^{2}})^{\frac{1}{2}}$

and $e^{\int_{0}^{\infty} \frac{tan^{-1}(\frac{-60\lambda O}{\epsilon-1})}{2}}$

Then $\sqrt{\epsilon_{c}-1} = TINY(\cos e^{\int_{0}^{\infty} \frac{tan^{-1}(\frac{-60\lambda O}{\epsilon-1})}{2}})$
 $\frac{\sqrt{\epsilon_{c}-1}}{\epsilon_{c}} = TINY(\cos e^{\int_{0}^{\infty} \frac{tan^{-1}(\frac{-60\lambda O}{\epsilon-1})}{2}})$

Therefore, for vertical polarization, l = 1, 2

$$f(hi) = 1 + j \left[\frac{2\pi hi}{\lambda} \cdot TINY \cdot \frac{\epsilon \cdot \cos \alpha - W \cdot SIN\alpha + j \left(W \cdot \cos \alpha + \epsilon \cdot SIN\alpha\right)}{\lambda} \right]$$
REAL PART = 1 + $\frac{2\pi hi}{\lambda}$ $TINY \cdot \frac{W \cdot \cos \alpha + \epsilon \cdot SIN\alpha}{\epsilon^2 + W^2}$

and for horizontal polarization, i = 1, 2REAL PART = $\int -\frac{2\pi h_i}{\lambda}$. TINY · SIN &

For both cases

For vertical polarization

$$S = \left(\frac{2\pi ka}{\lambda}\right)^{\frac{2}{3}} \frac{\hat{\epsilon}_{c^{-1}}}{\hat{\epsilon}_{c^{2}}}$$

k = index of refraction

Let
$$H = \left(\frac{2\pi \kappa a}{\lambda}\right)^{\frac{2}{3}}$$

Then $S = H \cdot \left[\frac{(\epsilon - i) - j W}{(\epsilon^2 - w^2) - j(2\epsilon W)}\right]$

Then
$$S = H \cdot \left[\frac{(\epsilon - i) - jW}{F - jG} \right]$$

=
$$H \cdot \left[\frac{(E-I)F + GW}{F^2 + G^2} + J \cdot \frac{(E-I)G - FW}{F^2 + G^2} \right]$$

Let
$$CBM = H \cdot \left[\frac{(e-i)F + GW}{F^2 + G^2} \right]$$

and
$$CAW = H \cdot \left[\frac{(E-I)G - FW}{F^2 + G^2} \right]$$

For horizontal polarization

$$S = H(\hat{\epsilon}_{e^{-1}}) = H\left[(e^{-1}) - jW\right]$$

Let CEM = H
$$(\epsilon - i)$$

For both cases

$$OC = CEM + j CAW$$

distance factor

(3)
$$\mathcal{J} = \left(\frac{2\pi}{1 - \kappa^2 a^2}\right)^3 \cdot o$$

d = distance between antennas

and for n = 4, 5, 6,150 = 4.382 F, = 1.856 = 3.245= 4.3823.245

$$_{,\infty}$$
 = $\frac{1}{2}$ $\left[3\pi(n+\frac{1}{2})\right]^{\frac{2}{3}}\left(\frac{1}{2}-j\frac{\sqrt{3}}{2}\right)$

Let TNR = real part of TNI = imaginary part of Th, 00)E,

Tigo = TNR + j TNI

Since of = CEM + i CAW

Let SMALL =
$$\left(\sqrt{CEM^2 + CAW^2}\right)^{-\frac{1}{2}} = \left(\sqrt{CEM^2 + CAW^2}\right)^{-\frac{1}{2}} = -\frac{1}{2} \frac{tan^{-1} \left(\frac{CAW}{CEM}\right)}{tan^{-1} \left(\frac{CAW}{CEM}\right)}$$
and $\chi = \frac{tan^{-1} \left(\frac{CAW}{CEM}\right)}{2}$

- SMALL (cas & -j SINY)

(VCEM 2 + CAW 2 O NIN 3 ton (CAW)

Let CALL = $\left(\sqrt{CEM^2 + CAW^2}\right)^{-\frac{1}{2}}$

and $\phi = \frac{3}{2} tan^{-1} \left(\frac{CAW}{CEM} \right)$

Then S-= CALL (COS \$-j SW\$)

- 12 7,8 S-3 = -3 [TNR+jTNI]. [CALL (COS \$ - jSIN\$)]

- F CALL [TNR COS \$ + TNI SIN\$ + j (TNI COS \$ -TNR SIN\$)]

Then, referring to equation (4)

 $\kappa = \text{TNR} + j \text{TNI} - \text{SMAIL} (\cos \gamma - j \sin \gamma)$

- 3 CALL [THR COS \$ + THI SING + j (THI COS \$ -THR SIND)]

and

Let the real part of 7n imaginary part of Th= TNI + SMALL SIN & - 3 CALL (TWI COS \$\phi - TWR SW\$) real part of in = INR - SMALL COS $\sqrt{-\frac{2}{3}}$ CALL (THR COS \$ + THI SIN \$)

and the imaginary part of $\mathcal{T}_{\eta} = IP$

Referring to equation (1)

$$\frac{e^{-j \pi s}}{e} = \frac{e^{-j (RP + j IP) s}}{(cen + j caw) + (2RP + j 2IP)}$$

Let
$$q = IPS$$
, $\beta = RP$.

CEM + 2 RP, B = CAW + 2 IP
$$\frac{e^{-jT_n x}}{e^{-jP}} = \frac{e^{-jP}}{A + jB}$$

Then

$$= e^{\alpha} \left[A \cos \beta - B \sin \beta - i \left(B \cos \beta + A \sin \beta \right) \right]$$

$$A^2 + B^2$$

REAL PART =
$$e^{\alpha}(A\cos\beta - B\sin\beta)$$

IMAGINARY PART =
$$-e^{\alpha}(B\cos\beta + A\sin\beta)$$

 $A^2 + B^2$

Hence
$$\frac{7}{s} = \frac{e^{-j \pi S}}{\frac{6+27n}{n}} = \sum_{n=1}^{7} \frac{\text{REAL PART}}{\frac{n}{n}} + j = \sum_{n=1}^{7} \frac{\text{IMAGINARY PART}}{\frac{n}{n}} = \left(\sum_{n=1}^{2} \frac{REAL PART}{\frac{n}{n}}\right)^{2} + \left(\sum_{n=1}^{7} \frac{\text{IMAGINARY PART}}{\frac{n}{n}}\right)^{2}$$
(5) $\frac{e^{-j \pi S}}{\frac{n}{s}} = \left(\sum_{n=1}^{2} \frac{REAL PART}{\frac{n}{n}}\right)^{2} + \left(\sum_{n=1}^{7} \frac{\text{IMAGINARY PART}}{\frac{n}{n}}\right)^{2}$

Now the results of equations (2), (3), and (5) can be incorporated into equation (1).

BREMMER SERIES

OPERATION OF PROGRAM

- Follow steps 1 through 3 for the operation of the Free Space Loss program.
- 'n Commands will be of increments. mmands will be typed out as in the previous programs. Remember to cautious and type in everything in floating point except the number increments. Also type everything in the same units of distance.
- ω This program is in two part numbers will be typed out. of C, $|f(h_1)|^2$, $|f(h_2)|^2$ in two parts. and These are real part of o, imaginary part and (277 K2 22)
- + Load address 200 and start. Depress continue. The second half of the tape will load.
- 5. The command "Type the above the five numbers typed from typed from Part numbers" will be typed. H This refers t o
- 9 commands will be typed, so enter the appropriate data
- ~ "Bremmer (db)" = the answer is typed and the next distance s, requested.

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EXPLANATION OF PROGRAM

The Bremmer Series Program begins by accepting

 $AMBA = \lambda$, wave length

REFR = k, index of refraction

A = a, radius of body

H1, H2 = h, h2, heights of antennas

 $E = \epsilon$, dielectric coefficient

 $S = \mathbf{\mathcal{C}}$, conductivity

ing order: From the beginning to label 33 the program computes independent of polarization, distance, and n. They are, values that in the follow-

At label 33 the operator, depending horizontal or vertical polarization. depending on his requirements, can select

For vertical polarization the program goes to label 3^{4} . part and imaginary part of \mathcal{O} , CEM and CAW respectively, and Y, factors in the calculation of $f(h_{i})$, are compute tal polarization the program goes to label 35 where similar reasons. made. where similar calculations along with X Here the real For horizon-

The program then proceeds to label 19 where the previous calculations are used to compute the real part of $f(h_i)$, the imaginary part of $f(h_i)$, the real part of $f(h_2)$, the imaginary part of $f(h_2)$, $|f(h_i)|^2$, the imaginary part of $f(h_2)$, $|f(h_i)|^2$, and $(2 \frac{\pi}{2}) \frac{1}{2} \frac{2}{3} \frac{1}{3}$

The program then types out the values for CEM, CAW, $|f(h_1)|^2$, $|f(h_2)|^2$, $(2K_{\kappa^2}a^2)^3$ and stops. This is the end of the first part of the program.

Part 2 begins by requesting the operator to type the numbers typed at the end of Part 1. The program then goes into a "DO LOOP" which depends on the number of distances that are requested. The particular distance is then entered and

is computed.

The program then enters a nested "DO LOOP" in order to compute $\frac{7}{\sqrt{-i}}$

SUMR and RUMI will eventually be the real and imaginary parts of the formula.

During the first time through this nested "DO LOOP", TNR = .928 then the program goes to label 17 where TNI = $\sqrt{3}$ (.928). These values correspond to the real and imaginary parts of 7

The program then computes the real and imaginary parts of

denoted by REAL and EMAG respectively.

Next the program uses these results to compute

$$\frac{e^{-jT_{i,\infty}J}}{cC+2T_{i,\infty}} = \frac{e^{-jTNI}}{(cEM+2TNR)+j(CAW+2TNI)}$$

Then the real part is added to SUMR while the imaginary part is added to

During the second and third time through the "DO LOOP" INR = 1.6225 and 2.191 respectively, and the real and imaginary values are computed and added to SUMR and RUMI sespectively. However, for the fourth through the seventh times, the equation

INR =
$$\frac{1}{7} (3\pi(n+\frac{1}{7}))^{\frac{3}{3}}$$
 $n = 4, 5, 6, 7$

loop. EN corresponds to n + 1/4 and is The program then goes to 17 where it proceeds as before. incremented by one each pass through the

After seven passes through the nested "DO LOOP". The program computes

$$\left|\sum_{S'+2T_n}^{2}\right|^2 = \text{SUMR}^2 + \text{RUMI}^2$$

BREW = CONST * SBA * ABS * VALUE is computed, and this result is then expressed in decibels by the formula

4.34294*LOGF (BREM) which corresponds to 10 LOG₁₀ (BREM)

which corresponds to the general equation for the Bremmer Series expressed in decibels
$$L_{\rm B}({\rm db}) = 10 \log_{10} \left(\frac{|\hat{\mathbf{E}}|^2}{|\mathbf{E}|^2} \right) = 10 \log_{10} \left[8 \hbar f \cdot |f(h_1)|^2 |f(h_2)|^2 \right] \frac{e^{-j \frac{\pi}{h_1} f}}{\sqrt{f + 2h_1}} \Big]$$